## FLAP SYSTEM - INSPECTION/CHECK

# 1. General

A. This section has the inspections and checks necessary to keep the flap system in a serviceable condition. **TASK 27-50-00-220** 

## 2. Flap Actuator Mount Bracket Detailed Inspection

- A. General
  - (1) This task gives the procedures to do a detailed inspection of the flap actuator mount bracket.
- B. Special Tools
  - (1) None
- C. Access
  - (1) Open (unzip) the fabric headliner (passenger) or remove the hard shelled headliner (cargo) to get access to the flap actuator mount bracket. Refer to Chapter 25, Cabin Upholstery Maintenance Practices.
- D. Do a Detailed Inspection of the Flap Actuator Mount Brackets.

WARNING: If cracks are found in the support structure, reinforce or replace the structure as necessary. Stop drilling cracks is not sufficient; more reinforcement is necessary.

- (1) Examine the flap actuator support structure for corrosion, cracks, deformation, or other signs of damage.
- E. Restore Access.
  - (1) Close (zip) the fabric headliner (passenger) or install the hard shelled headliner (cargo). Refer to Chapter 25, Cabin Upholstery Maintenance Practices.

#### END OF TASK TASK 27-50-00-221

### 3. Flap Bellcrank Detailed Inspection

- A. General
  - (1) This task gives the procedures to do a detailed inspection of the flap bellcranks.
- B. Special Tools
  - (1) None
- C. Access.
  - (1) Remove the necessary panels and covers to get access to the flap components on both wings. Refer to Flap Rigging Guide Adjustment/Test, Figure 501.
- D. Do a detailed inspection of the flap bellcranks.
  - (1) Clean and lubricate the flap bellcranks, interconnect rods, and pushrods. Refer to Chapter 12, Flight Controls Servicing.

CAUTION: Bellcrank supports are subject to high loads. Close inspection of the supports and the adjacent structure is mandatory.

- (2) Examine the flap bellcranks, bellcrank tubes, bearing, and bushings for corrosion, cracks, condition, deformations, signs of damage, and security of installation.
- (3) Examine the bellcrank supports for corrosion, cracks, condition, buckling, and security of installation.
- (4) Examine the tube ends for interference with the adjacent structure.
- E. Restore Access.
  - (1) Install the applicable panels and covers that were removed to get access to the flap components on both wings. Refer to Flap Rigging Guide Adjustment/Test, Figure 501.

### END OF TASK TASK 27-50-00-720

# 4. Flap System Functional Check

- A. General
  - (1) This task gives the procedures to do a functional check of the flap system.
- B. Special Tools

- (1) Cable Tensiometer
- (2) Inclinometer
- (3) External Electrical Power Unit
- (4) Torque Wrench
- C. Access
  - (1) Remove the necessary panels and covers to get access to the flap components on both wings. Refer to Flap Rigging Guide Adjustment/Test, Figure 501.
  - (2) Open (unzip) the fabric headliner (passenger) or remove the hard shelled headliner (cargo) to get access to the flap actuator and the wing-to-wing interconnect rod. Refer to Chapter 25, Cabin Upholstery Maintenance Practices.
- D. Complete a Functional Check of the Flap System.
  - (1) Examine the flap control lever and pointer for security of installation, travel, and signs of damage.
  - (2) Examine the flaps for loose rivets, cracks, condition, and security of installation.
  - (3) Examine the flap cable runs for interference with the structure, correct routing, frozen pulleys, fraying, twisting, and corrosion.
    - (a) Look for interference with the adjacent structure, equipment, wiring, plumbing, and other controls.
  - (4) Move a cloth along the full length of the flap cables to examine for broken wires.
    - (a) If snags are found or you think that there are broken wires, Refer to Chapter 20, Control Cable and Corrosion Limitations Maintenance Practices.
  - (5) Examine the pulleys, attach brackets, and guard pins for condition, wear, corrosion, and security.
  - (6) Turn the pulleys with your hand to make sure that there is freedom of movement, and to keep even wear on the pulleys.
  - (7) Examine the cable attachment brackets on each flap for condition, corrosion, security, and correct attachment of the cable to the bracket.
  - (8) Examine the motors and the transmission for condition, wear, corrosion, and security.
  - (9) Do the Flap Component Inspection. Refer to Flap Rigging Guide Adjustment/Test.
- E. Do a Travel and Cable Tension Check.
  - (1) To examine the flap travel and cable tensions, do the Operational Check of the flaps. Refer to Flap Rigging Guide Adjustment/Test.
- F. Do a Standby Flap Motor Operational Check (Refer to Figure 601).
  - CAUTION: You must set the NORMAL/STBY switch to STBY before you operate the standby UP/DOWN switch. Since the standby flap system bypasses the limit function of the flap switch actuator, you must stop the operation of the standby UP/DOWN switch before the flaps reach their limits. This will help prevent overloading and damage to the flap system.
  - (1) For Airplanes 20800224 and On and 208B0327 and On, and airplanes that incorporate SK208-119A, break the frangble copper wire on the UP/DOWN switch guard and the NORMAL/STBY switch guard.
  - (2) Set the battery switch to ON.

WARNING: Before you move the flaps, make sure that the area around the flaps is clear. This will prevent injuries to personnel and damage to the equipment and the flaps.

- (3) Use the flap control lever in the control pedestal to move the flaps to the 10 degree position.
- (4) Open the NORMAL/STBY switch guard.
- (5) Set the NORMAL STBY switch to STBY.
- (6) Move the flaps to the 20 degree position with the standby UP/DOWN switch.
- (7) Move the flaps to the 10 degree position with the standby UP/DOWN switch.
- (8) Close the NORMAL/STBY switch guard to set the NORMAL/STBY switch to NORMAL.
- (9) Move the flaps to the UP position with the flap control lever.
- (10) Set the battery switch to OFF.
- (11) For Airplanes 20800224 and On and 208B0327 and On, and airplanes that incorporate SK208-119A, use frangible

copper wire to safety the NORMAL/STBY switch guard and the UP/DOWN switch guard in the closed position.

- (12) Make sure that all rod end inspection holes are covered.
- (13) Make sure that the rod ends are positioned so maximum rotational freedom is available to each rod (so rod housings are perpendicular to attaching bolts).
- (14) Remove the inclinometers from left and the right flaps.
- (15) Make sure that the necessary flap system components are secure, torqued, and safety wired.
- (16) Put the External Power Switch to OFF.
- (17) Remove external electrical power from the airplane.

### G. Restore Access.

- (1) Close (zip) the fabric headliner (passenger) or install the hard shelled headliner (cargo). Refer to Chapter 25, Cabin Upholstery Maintenance Practices.
- (2) Install the applicable panels and covers that were removed to get access to the flap components on both wings. Refer to Flap Rigging Guide Adjustment/Test, Figure 501.

# END OF TASK

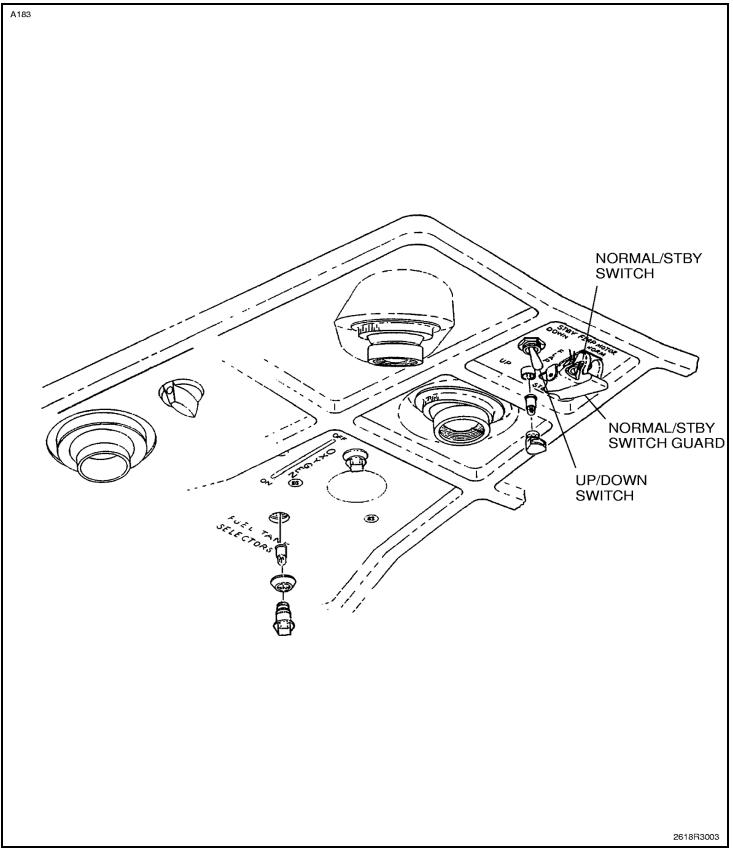
# TASK 27-50-00-640

# 5. Flap Tracks and Rollers Lubrication

- A. General
  - (1) This task provides the procedures to perform a lubrication of the flap tracks and rollers.
- B. Tools and Equipment
  - (1) External Electrical Power Unit, 28 VDC.
  - (2) Dry Solid Film Lubricant (MIL-L-23398D)
- C. Access
  - (1) None
- D. Complete a Lubrication of the Flap Tracks and Rollers.
  - (1) Connect the external electrical power unit to the airplane.
  - (2) Set the External Power Switch to the BUS position.
  - (3) Set the Battery Switch to the ON position.
  - (4) Extend / retract the flaps as necessary to get access to the tacks and rollers.
  - (5) Wipe the flap tracks and the rollers clean and examine for corrosion.
  - (6) Lubricate the flap tracks and the rollers with dry solid film lubricant (MIL-L-23398D).
  - (7) Wipe off unwanted spray.
  - (8) Fully retract the flaps.
  - (9) Set the Battery Switch to the OFF position.
  - (10) Set the External Power Switch to the OFF position.
  - (11) Remove the external electrical power unit from the airplane.
- E. Restore Access

#### (1) None END OF TASK





Model 208/208B Maintenance Manual (Rev 33) 27-50-00-6 (Rev 24)



